Chapter 5. Incompressible Flow in Pipes and Channels

Shear Stress and Skin Friction in Pipes (전단응력 및 표면마찰)

*** Shear-stress distribution**

For fully developed flow, $V_b = V_a \& \beta_b = \beta_a$ ∴ $\Sigma F = 0$ ← from Eq. (4.51): $\Sigma F = m(\beta_b \overline{V}_b - \beta_a \overline{V}_a)$

From Eq. (4.52),
$$
\sum F = p_a S_a - p_b S_b + F_w - F_g = 0
$$

= $-F_s$

$$
\sum F = \pi r^2 p - \pi r^2 (p + dp) - (2\pi r dL)\tau = 0
$$
\n
$$
\frac{dp}{dL} + \frac{2\tau}{r} = 0 \qquad \text{--- Eq. (5.1)}
$$
\n
$$
\downarrow \qquad \qquad \downarrow \qquad \qquad \downarrow \qquad \pi^2 dL \equiv \text{L} + \text{H}
$$
\n
$$
\downarrow \qquad \qquad \downarrow
$$
\n
$$
r \downarrow \qquad \downarrow \equiv \text{H} \equiv \text{H} \quad (\text{H} \trianglelefteq \text{H}) \equiv \text{H} \equiv \text{H} \quad (\text{H} \trianglelefteq \text{H}) \equiv \text{H} \equiv \text{H} \quad (\text{H} \trianglelefteq \text{H}) \equiv \text{H} \quad (\text{H} \trianglelefte
$$

$$
\therefore \frac{dp}{dL} + \frac{2\tau_w}{r_w} = 0 \qquad \text{--- Eq. (5.2)}
$$

Eq. (5.2)에서 Eq. (5.1)을 빼 면,

$$
\frac{\tau_w}{r_w} = \frac{\tau}{r} \quad \text{or} \quad \boxed{\tau = \left(\frac{\tau_w}{r_w}\right)r}
$$

*** Relatio n between skin friction & wall she a r**

펌프에 의한 일이 없고 마찰을 고려할 경우의 Bernoulli 방정식은

$$
\frac{p_a}{\rho} + gZ_a + \frac{\alpha_a \overline{V}_a^2}{2} = \frac{p_b}{\rho} + gZ_b + \frac{\alpha_b \overline{V}_b^2}{2} + h_f \qquad \text{--- Eq. (4.71)}
$$

일반적으로 p_a > p_b 이므로 p_b = p_a - Δp 로 표시할 수 있고 fully developed flow인 수평관을 대상으로 하며 마찰은 유체와 관벽 사이의 skin friction *hfs* 만 존재하므로 $V_b = V_a, \; \alpha_b = \alpha_a, \; Z_b = Z_a, \; \& \; \Delta p = \Delta p_s$ (압력강하는 표면마찰에 의한 것이므로)

이 경 우 Bernoulli 식은

$$
\frac{p_a}{\rho} = \frac{p_a - \Delta p_s}{\rho} + h_{fs} \quad \cong , \quad \frac{\Delta p_s}{\rho} = h_{fs} \qquad \text{--- Eq. (5.4)}
$$

From Eq. (5.2),
$$
\frac{-\Delta p_s}{L} + \frac{2\tau_w}{r_w} = 0
$$

$$
\therefore h_{fs} = \frac{2}{\rho} \frac{\tau_w}{r_w} L = \frac{4}{\rho} \frac{\tau_w}{D} L \qquad \angle
$$

*** F riction factor (**마찰계수**),** *f*

Å 여기서 정의하는 마찰계수 *f* 는 *Fanning friction factor*

또다른 마찰계수로 *Blasius* or *Darcy friction factor*가 있 는 데 이 는 ⁴*f* 에 해 당

$$
f \equiv \frac{\tau_w}{\rho \overline{V}^2 / 2} = \frac{2\tau_w}{\rho \overline{V}^2}
$$
 -- Eq. (5.6)
\nwall shear stress
\ndensity × velocity head \approx , $\frac{(\overline{C} \overline{C}) \overline{C} \overline{C} \overline{C} \overline{C} \overline{C} \overline{C})}{(\overline{C} \overline{C}) \overline{C} \overline{C} \overline{C} \overline{C} \overline{C} \overline{C})}$

skin friction *hfs* 와 friction factor *f* 와의 관계:

$$
h_{fs} = \frac{2 \tau_w}{\rho} L = \frac{\Delta p_s}{\rho} = 4f \frac{L \overline{V}^2}{D \overline{2}}
$$
 --- Eq. (5.7)

$$
\therefore f = \frac{\Delta p_s D}{2L\rho \overline{V}^2} \quad \text{or} \quad \frac{\Delta p_s}{L} = \frac{2f\rho \overline{V}^2}{D}
$$
 --- Eqs. (5.8)-(5.9)

*** Flow in noncircular channels**

In evaluating the diameter in noncircular channels, an equivalent diameter (등가지름) *Deq* is used.

$$
D_{eq} = 4r_H
$$
 \rightarrow r_H : hydraulic radius (수력학적 반지를)

 $r_H \equiv \frac{S}{s}$ *S* : cross-sectional area of channel *Lp* : wetted perimeter

단면이 원형이 아닌 관의 경우 Reynolds number *Re* 또는 friction factor *f* 등의 계산시에 D 대신 $D_{_{eq}}$ 혹은 r 대신 2 $r_{_{H}}$ 를 대입하여 계산 가능함을 의미.

Laminar Flow in Pipes and Channels

*** L aminar flow of Newtonian fluids**

원형 단면을 갖는 흐름을 대상, 속도분포는 centerline에 대해 대칭

u depends only on *r*

적분하면 (경계조건: u = 0 at r = $r_{_W})$

$$
\int_0^u du = -\frac{\tau_w}{r_w \mu} \int_{r_w}^r r dr \longrightarrow u = \frac{\tau_w}{2r_w \mu} \left(r_w^2 - r^2 \right) \qquad \text{--- Eq. (5.15)}
$$

$$
u_{\text{max}} = \frac{\tau_w r_w}{2\mu} \quad \text{(at } r = 0)
$$

$$
\therefore \frac{u}{u_{\text{max}}} = 1 - \left(\frac{r}{r_w} \right)^2 \qquad \text{--- Eq. (5.17)}
$$

Average velocity

$$
\overline{V} = \frac{1}{S} \int u dS \qquad --\text{ Eq. (4.11)}
$$
\n
$$
dS = 2\pi r d
$$
\n
$$
\text{Eq. (5.15) } \left\{ \frac{d}{dt} \right\} = \frac{dS}{dt} = 2\pi r d
$$
\n
$$
\overline{V} = \frac{\tau_w}{r_w^3 \mu} \int_0^{r_w} \left(r_w^2 - r^2 \right) r dr = \frac{\tau_w r_w}{4\mu} \qquad --\text{ Eq. (5.18)}
$$
\n
$$
0 \text{ A} \equiv u_{\text{max}} = \frac{\tau_w r_w}{2\mu} \text{ H } \overline{u} \text{ of } \overline{u}_{\text{max}} = 0.5 \qquad --\text{ Eq. (5.19)}
$$

 \rightarrow In Laminar flow,

 $\emph{Kinetic energy correction factor},$ α = 2.0 $\,$ \leftarrow Eq. (4.70)에 (5.15)와 (5.18)을 대입해 계산 *Momentum correction factor*, 3 $\beta = \frac{4}{3}$ ← Eq. (4.50)에 (5.15)와 (5.18)을 대입해 계산

Hagen-Poiseuille equ ation

Eq. (5.7)과 Eq. (5.18)을 이용하여 τ_w 대신 보다 실제적인 Δp_s로 변환하면,

$$
\overline{V} = \frac{\Delta p_s D^2}{32L\mu} \quad \text{or} \quad \boxed{\Delta p_s = \frac{32L\overline{V}\mu}{D^2}} \quad \text{--- Eq. (5.20)}
$$

여기서
$$
q = \frac{\pi D^2}{4} \overline{V}
$$
 이르로 q 외 4p_s 측정으로부터 점도 계산 가는:

$$
\mu = \frac{\pi \Delta p_s D^4}{128Lq}
$$
: Hagen-Poiseuille equation

또한 Eq.(5.7)에서 Δ $p_{_S}$ = 4 $\tau_{_W}$ /(*DL*)이므로,

$$
\tau_w = \frac{8\overline{V}\mu}{D} \qquad \qquad \text{--- Eq. (5.21)}
$$

Eq. (5.21)을 Eq. (5.7)에 대입하면 f 와 Re 사이의 관계가 유도됨:

$$
f = \frac{16\mu}{D\overline{V}\rho} = \frac{16}{\text{Re}}
$$
 -- Eq. (5.22)

*** L aminar flow of non-Newtonian liquids**

- Power l aw fluids

$$
\tau = -K \frac{du^n}{dr}
$$

반지름 *r* 에 따 른 velocity profile:

$$
u = \left(\frac{\tau_w}{r_w K}\right)^{1/n} \frac{r_w^{1+1/n} - r^{1+1/n}}{1+1/n}
$$

Fig. 5.4. Velocity profiles in the laminar flow of Newtonian and non-Newtonianliquids.

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- Some non-Newtonian mixtures at high shear violate the zero-velocity (**no-slip**) b. c. ex) multiphase fluids (suspensions, fiber-filled polymers) → "slip" at the wall

Turbulent Flow in Pipes and Channels

 \rightarrow Most of the kinetic-energy content of the eddies lies in the buffer zone.

*** Velocity distribution for turbulent flow**

In terms of dimensionless parameters

$$
u^* = \overline{V} \sqrt{\frac{f}{2}} = \sqrt{\frac{\tau_w}{\rho}}
$$
: friction velocity
\n
$$
u^+ = \frac{u}{u^*}
$$
: velocity quotient (exists)
\n
$$
y^+ = \frac{yu^* \rho}{\mu} = \frac{y}{\mu} \sqrt{\tau_w \rho}
$$
: distance (exists) *y*: distance from tube wall
\n
$$
\frac{u^*}{\mu} = \frac{yu^* \rho}{\mu} = \frac{y}{\mu} \sqrt{\tau_w \rho}
$$

$$
\therefore \text{distance (} \neq \overline{\lambda} \neq 0 \text{)} \qquad \text{y: distance from tube wall}
$$
\n
$$
\therefore r_w = r + y
$$

*** Universal velocity distribution equations**

i) viscous sublayer: $u^+ = y^+$ ii) buffer layer: $u^+ = 5.00 \ln y^+ - 3.05$ iii) turbulent core: $u^+ = 2.5 \ln y^+ + 5.5$ u^+ = 5.00 ln y⁺ $u^+ = 2.5 \ln y^+$ → intersection으로부터 for viscous sublayer for buffer zone for turbulent corefor turbulent core $<$ 5 + *y* $5 < y^+ < 30$ $y^+ > 30$

→ Re>10,000 이상에서 적용 가능

*** Relations between maximum velocity** max*u* **& average velocity** *V*

For laminar flow, $\overline{V}/u_{\rm max}^{\,}$ is exactly 0.5. \leftarrow from Eq. (5.19)

When laminar flo w changes to turbulent, the ratio V/u_{max} changes rapidly from 0.5 to about 0.7,

 $\&$ increases gradually to 0.87 when Re=10⁶.

*** Effect of rou g h ness**

Rough pipe \rightarrow larger friction factor $\leftarrow k$: roughness parameter *k/D* : relative roughness For laminar flow, roughness has no effect on *f* unless *k* is so large. ∴ $f = ft'n$ of Re & k/D

Types of roughnes s

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*** Friction factor chart**

Friction factor plot for circular pipes (log-log plot)

*** Friction factor for smooth tube**

$$
f = 0.046 \text{Re}^{-0.2} \quad \text{for } 50,000 < \text{Re} < 10^6
$$
\n
$$
f = 0.014 + \frac{0.125}{\text{Re}^{0.32}} \quad \text{for } 3,000 < \text{Re} < 3 \times 10^6
$$
\n(wide range)

*** Non-Newtonian fluids**

*** Drag reduction**

Dilute polymer solutions in water

- \rightarrow drag reduction in turbulent flow
- Application: fire hose (a few ppm of PEO in water can double the capacity of a fire hose)

*** Friction loss from sudden expansion**

2 $e = K_e \frac{\overline{V}_a^2}{2}$ $h_{fo} = K_e \frac{V_a^2}{V_a}$ $\left\{ \begin{array}{l} V_a : \text{average velocity of smaller or upstream section} \right\}$ *K*_e: expansion loss coefficient *Va*

Ke can be calculated theoretically from the momentum balance equation (4.51) and the Bernoulli equation (4.71).

$$
K_e = \left(1 - \frac{S_a}{S_b}\right)^2
$$
 for turbulent flow $(\alpha \approx 1 \& \beta \approx 1)$

Laminar flow인 경우에는 α = 2 & β = 4/3을 사용하면 *K_e* 를 구할 수 있다.

*** Friction loss from sudden contraction**

2 $f_c = K_c \frac{\overline{V_b}^2}{2}$ $h_{fo} = K_c \frac{V_b^2}{V_b}$ [*V_b*: average velocity of smaller or downstream section) K_c : contraction loss coefficient V_b

 $K_c < 0.1$ h_c < 0.1 for laminar flow $\rightarrow h_{fc}$ is negligible. ⎟ ⎟ ⎠ ⎞ ⎝ $= 0.4 \left(1 - \frac{1}{2} \right)$ *a* $b_c = 0.4 \left(1 - \frac{b}{S_a}\right)$ $K_c = 0.4 \left(1 - \frac{S_b}{S_c}\right)$ for turbulent flow (empirical equation)

*** F riction loss from fittings**

$$
h_{ff} = K_f \frac{\overline{V_a}^2}{2} \qquad \begin{cases} \overline{V_a} : \text{average velocity in pipe leading to fitting} \\ K_f : \text{fitting loss coefficient} \end{cases}
$$

Table 5.1 \rightarrow Loss coefficients for standard pipe fittings

*** Minimizing expansion and contraction losses**

. **Contraction loss** can be nearly eliminated by r educing the cross section gradually.

 $K_c \approx 0.05$

In this case, separation & *vena contracta* do not occur.

. **Expansion loss** can also be minimized by enlarging the cross section gradually

To minimize expansion loss, the angle between the diverging walls of a conical expander must be less than 7°.

For <u>angles $> 35^{\circ}$ </u> \rightarrow The loss through this expander can become greater than that through a s udden expansion.

Separation of boundary layer in diverging channel

*** Flow through parallel plates** (Prob. 5.1 & 5.3과 연 관)

In laminar flow between infinite parallel plates,

$$
p_a - p_b = \frac{12\mu VL}{b^2} \text{ } \cong \cong \text{ } \pm 0, \pm 1, u/u_{\text{max}}, \overline{V}/u_{\text{max}} \equiv \pm 1, \pm 1, \pm 2.
$$

(풀이) Force balance: 2*yWp a* $-2yWp_b = 2τLW$ \leftarrow from Eq. (4.52)

$$
\frac{p_a - p_b}{L} = \frac{\tau}{y} \qquad \qquad \frac{\tau}{L} = -\mu \frac{du}{dy}
$$

$$
\frac{(p_a - p_b)}{L\mu} \int_{b/2}^{y} y \, dy = -\int_0^u du
$$

적분하면,

$$
u = \frac{(p_a - p_b)}{2\mu L} \left(\left(\frac{b}{2}\right)^2 - y^2 \right) \xrightarrow{(u_{\text{max}}} \text{at } y = 0)
$$

$$
\therefore u_{\text{max}} = \frac{(p_a - p_b)}{2\mu L} \left(\frac{b}{2} \right)^2
$$

Related problems:

(Probs.) 5.4, 5.8, 5.10, 5.12, 5.13, 5.17, 5.20 and 5.21

